



## Modeling and Analysis of Structural Dynamics for a One-Tenth Scale Model NGST Sunshield

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### **Overview**



- Introduction
- Membrane Modeling
- Finite Element Model
- Analysis of Sunshield Dynamics
  - Modal Analysis
  - Impulse Test Simulation
- Comparison of Analysis and Experiment
- Closing Remarks



### Introduction



- The Next Generation Space Telescope (NGST) will require a lightweight, deployable sunshield to provide passive cooling for optical systems.
- NGST 'yardstick' concept sunshield characteristics:
  - Central support structure
  - Deployable support booms
  - Pretensioned, thin-film membranes
- Technology development needs:
  - Modeling and Analysis
    - High confidence, test validated structural models to predict potential disturbances to observatory due to sunshield dynamics
    - · Assessment of membrane wrinkling effects
  - Testing
    - Scale model testing methodologies
    - Advanced, non-contact instrumentation
- Analysis and ground testing of a one-tenth scale model of the NGST 'yardstick' concept sunshield is being carried out to develop and validate capabilities to predict and verify sunshield structural characteristics.



**NGST 'Yardstick' Concept** 

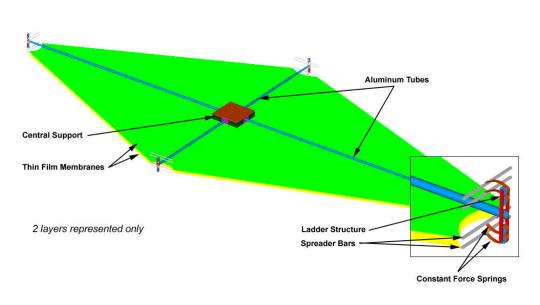


One-Tenth Scale Model
Sunshield Ground Test Article



### **One-Tenth Scale Model NGST Sunshield**





### **Test Article Characteristics:**

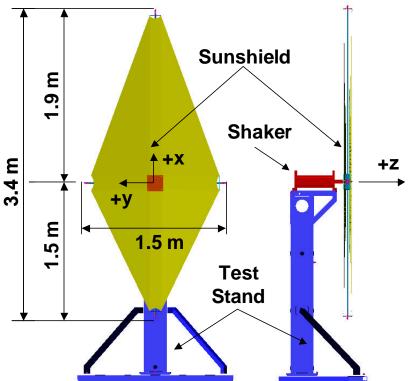
Overall size: 3.4 m x 1.52 m

Membranes: 4 layers of 13 micron (0.5 mil) Kapton

Support booms: Aluminum tubes

Membrane preload: 1.425 N (0.32 lbs) per corner

# Schematic of Test Setup with Sunshield in Short Side Down Orientation





## **Sunshield Dynamic Tests**



- Dynamic testing of the one-tenth scale model NGST sunshield in a vacuum environment was completed in August 2000 at NASA GSFC.
- Test objectives:
  - Determine the dynamic characteristics of the sunshield (frequencies, mode shapes, and damping) in order to provide data for model validation studies.
  - Verify the planned on-orbit test methodology for the Inflatable Sunshield In Space (ISIS) flight experiment.
  - Characterize the linearity of the system.
  - Characterize the influence of gravity.
- Types of tests:
  - Random excitation tests
  - Impulse excitation tests
  - Sine dwell tests
- Instrumentation:
  - Accelerometers
  - Force transducers
  - Laser vibrometer



## **Membrane Modeling**



- A challenging aspect of sunshield analysis is modeling the behavior of partially wrinkled thin-film membranes.
- Finite element modeling techniques for membranes:
  - Standard elements formulations:
    - Ignore membrane wrinkling
    - · Numerical problems may occur
  - Cable network method:
    - Approximate/engineering approach for modeling wrinkled membranes
    - Based on the established principal that load transfer in wrinkled regions takes place along wrinkle lines
  - Advanced techniques:
    - Use of tension field theory and special material models to account for wrinkling effects on membrane stress distributions.
    - 'Fine-scale' analysis of wrinkling enables the prediction both stress distribution and wrinkling details (wrinkle amplitude, wavelength, etc).
- The one-tenth scale model NGST sunshield finite element model utilized in the current study was developed using the cable network modeling approach.

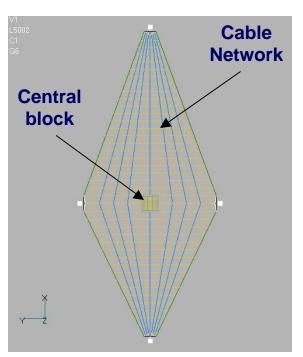


### **Finite Element Model**

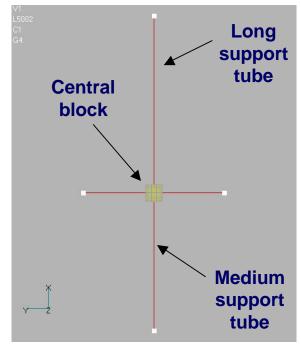


- Partially wrinkled, thin-film membranes modeled using 'cable network' technique
  - Developed by M. Mikulas/University of Colorado-Boulder
  - Implemented in UAI NASTRAN by S. Lienard at NASA GSFC
- Central block, support tubes, and tip hardware modeled using standard techniques
- Base-drive shaker support condition modeled (0.4 Hz rigid body translational mode of sunshield)

#### **Full FEM**

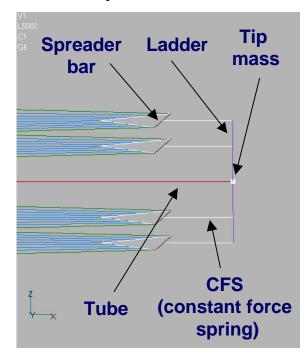


### **Support Structure**



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### **Tip Hardware**



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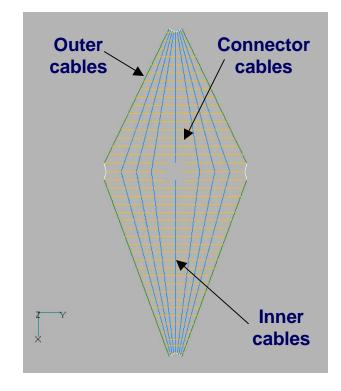


### **Membrane Cable Network**



- The wrinkle pattern in the longitudinal direction is dominant
  - 10 psi longitudinal stress
  - 5 psi lateral stress
- The membrane is meshed with a network of preloaded 'cables' mapped to the wrinkle pattern of the structure.
  - Longitudinal cables are oriented along the wrinkle pattern (load path)
  - Transverse cables act as a connection between cables and represent the mass distribution
  - Approximate representation of the load paths and mass distribution in the structure







## **Analysis of Sunshield Dynamics**



- Finite element analysis of the sunshield was completed using UAI NASTRAN (Version 20.1).
- Nonlinear static analysis
  - Preloads due to constant force springs
  - Gravity loading
  - Export stiffness matrix
- Modal analysis
  - Import stiffness matrix from nonlinear static analysis
  - Calculate frequencies and mode shapes
  - Calculate modal effective mass
- Other dynamic analyses
  - Frequency Response
  - Transient Response (Impulse test simulation)



## **Modal Analysis Results**



- UAI NASTRAN Normal Modes Analysis (SOL 3)
- Modal analysis completed for the following cases:
  - CFS preloads only
  - CFS preloads + Gravity loading with short side down orientation (-X direction)
  - CFS preloads + Gravity loading with sunshield in long side down orientation (+X direction)
- Effective mass (EFFM) calculations used to select 'significant' sunshield modes

Mode	No Gravity		Gravity Short Side Down		Gravity Long Side Down		Description
	Frequency (Hz)	EFFM (%)	Frequency (Hz)	EFFM (%)	Frequency (Hz)	EFFM (%)	
А	2.54	16.0	2.18	9.67	2.61	17.73	Long side of all membrane layers moving in-phase.
В	3.33	5.5	3.33	11.05	2.81	0.39	Short side of all membrane layers moving in-phase.
С	3.66	46.6	3.55	41.32	3.67	48.05	First long tube bending mode
D	5.51	4.9	4.71	3.64	4.79	4.31	Membrane/tube interaction mode
Е	5.71	12.8	5.54	14.54	5.55	13.27	First medium tube bending mode
F	6.63	5.9	6.43	9.49	6.32	8.91	Membrane/tube interaction mode

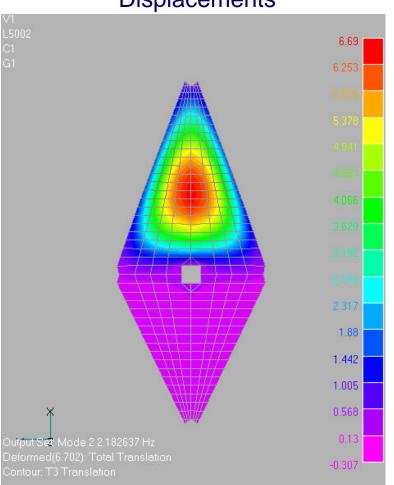


## **Modal Analysis Results – cont.**

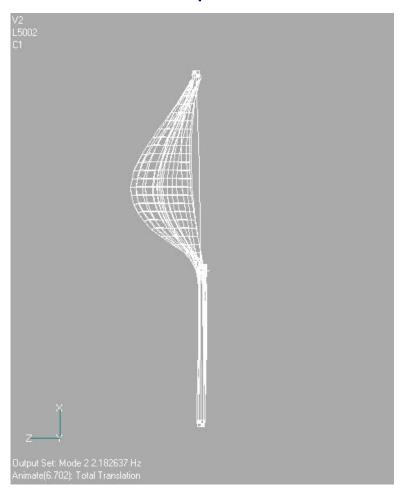


First Mode of Long Side of Membranes (F = 2.2 Hz)
Test article orientation = Short side down

## Contour Plot of Out-Of-Plane Displacements



### **Mode Shape Animation**



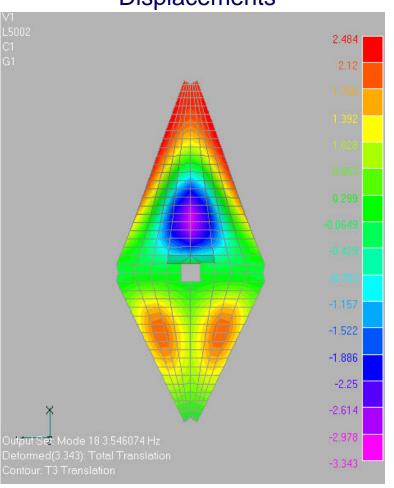


## Modal Analysis Results - cont.

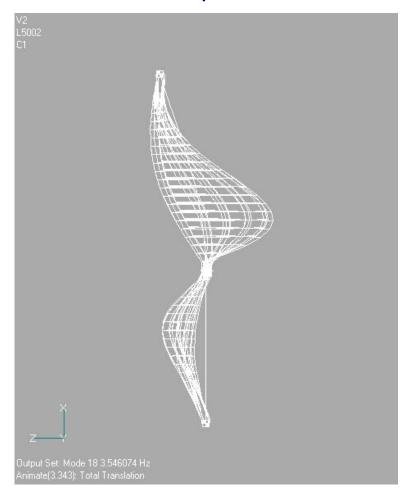


First Bending Mode of Long Tube (F = 3.5 Hz)
Test article orientation = Short side down

## Contour Plot of Out-Of-Plane Displacements



### **Mode Shape Animation**





### **Impulse Test Simulation**



- UAI NASTRAN Transient Response Analysis (SOL 12)
  - Modal Method (0 10 Hz frequency range)
  - Modal damping (using test-derived damping values)
  - Timestep = 0.0078125 s, Number of timesteps = 2048 , Total time = 16.0 s
- Loading:
  - Static: CFS preloads, gravity
  - Dynamic: force time history at shaker/sunshield interface from ground tests
- Time domain results recovered at key locations:
  - Accelerometers
  - Laser vibrometer measurement points
  - Force transducer at shaker/sunshield interface
- Frequency domain results calculated during post-processing using MATLAB
  - Power spectral densities
  - Transfer functions
    - Input = Applied load at shaker/sunshield interface
    - Output = predicted response at accelerometer, laser measurement locations



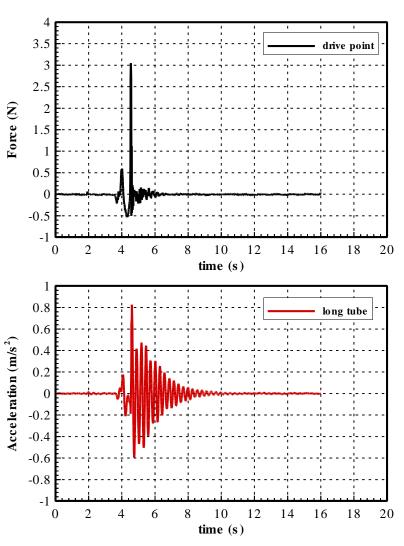
## **Impulse Test Simulation Results**

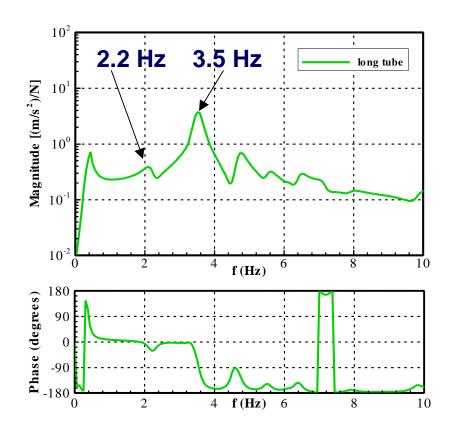






### **Frequency Domain Results**







## **Comparison of Analysis and Experiment**



### **Frequencies and Mode Shapes**

- Analysis / Test correlation performed using Dynaview software package
  - Analytical results for preload+gravity case
  - Test results from random excitation tests
  - Orthogonality calculation used to identify mode pairs
- In general, modes dominated by support tube response correlate better than modes dominated by membrane response.

# Comparison of Predicted/Measured Frequencies (Short Side Down Orientation)

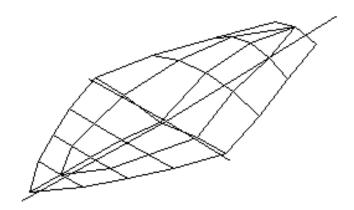
Predicted (Hz)	Measured (Hz)	Difference (%)	Description
2.18	3.00	-27.33	Membrane mode
3.55	3.48	+2.01	First bending mode of long tube
5.54	5.07	+9.27	First bending mode of medium tube
6.43	5.96	+7.89	Medium tube/short side of membranes

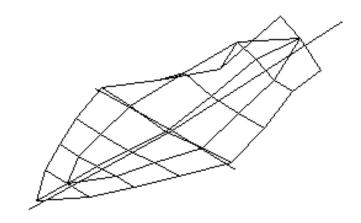


## **Mode Shape Correlation**



Analysis Model 2 2.18 Hz, 0.0000% Test Model 5 3.00 Hz, 4.7809%









### **Mode Shape Animations**

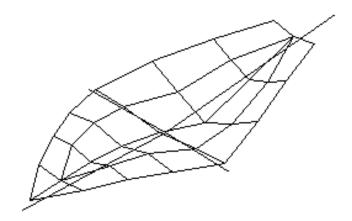
- Long side of membranes
- Long tube in-phase with membranes
- Frequency = 2.2 Hz (Analysis) / 3.0 Hz (Test)

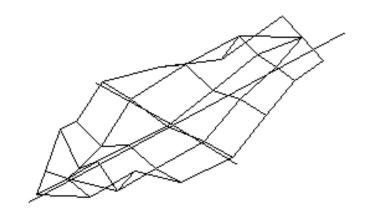


## **Mode Shape Correlation**



Analysis Model 4 3.55 Hz, 0.0000% Test Model 6 3.48 Hz, 5.2049%









### **Mode Shape Animations**

- First bending mode of long tube
- Long side of membranes out-of-phase with tube
- Frequency = 3.5 Hz (Analysis/Test)



## **Comparison of Analysis and Experiment**



### **Impulse Test Simulation – Support Tube Response**

Description	Predicted	Measured
(a) Time domain  Acceleration in z direction as a function of time at the tip of the long support tube.	0.8 0.6 0.9 0.2 0.2 0.4 0.6 0.8 0.6 0.8 0.1 0.2 0.2 0.4 0.6 0.8 0.8 0.9 0.9 0.9 0.9 0.9 0.9 0.9 0.9 0.9 0.9	1 0.8 long tube
<ul><li>(b) Frequency domain</li><li>Frequency response functions for tips of long and medium support tubes.</li><li>Input = drive point force</li><li>Output = acceleration of tube tips</li></ul>	10 <sup>2</sup> long tube ————————————————————————————————————	10 <sup>2</sup> long tube medium tube  10 <sup>3</sup> long tube medium tube  10 <sup>3</sup> long tube medium tube  10 <sup>4</sup> long tube medium tube  4 f(Hz) 6 8 10



## **Closing Remarks**



- Finite element analysis was used to predict the structural dynamic behavior of a one-tenth scale model of the NGST 'yardstick' concept sunshield.
  - Membranes modeled using approximate engineering technique that accounts for wrinkling effects.
  - Comparison of analytical predictions and test results showed good agreement for modes dominated by support tube motions, but only fair agreement for modes dominated by membrane response.

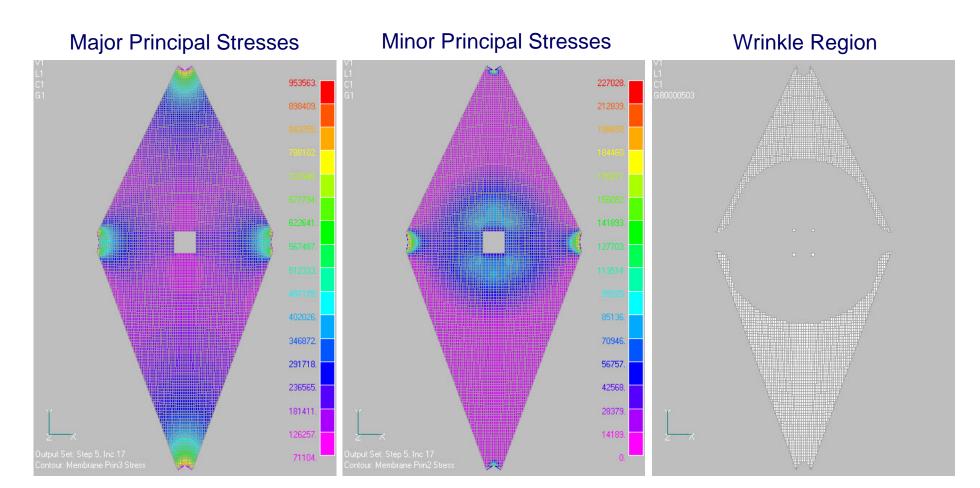
#### Current / Future Work:

- Development of a new membrane finite element model of the sunshield using "Iterative Membrane Properties (IMP)" method developed by Adler and Mikulas/U. Colorado-Boulder.
- Follow-on testing of one-tenth scale model sunshield.



# **Current Work**Wrinkled Membrane FEA





Membrane wrinkling effects modeled using IMP method developed by Adler and Mikulas/U. Colorado-Boulder